



AFBench:

A Large-scale Benchmark for Airfoil Design

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***Presenting**



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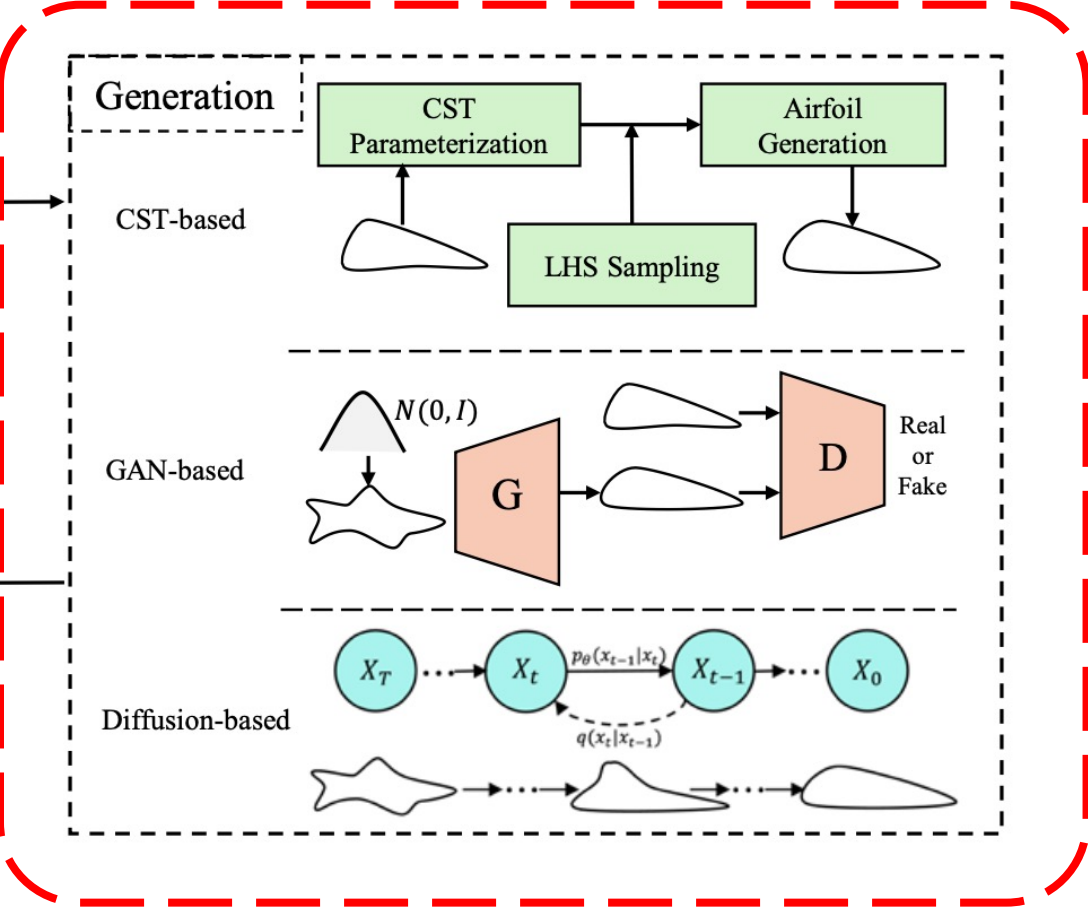
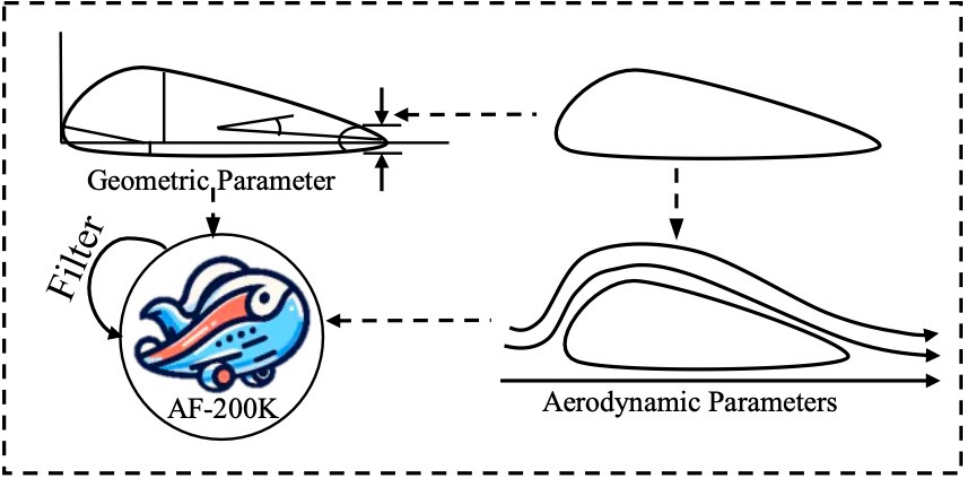
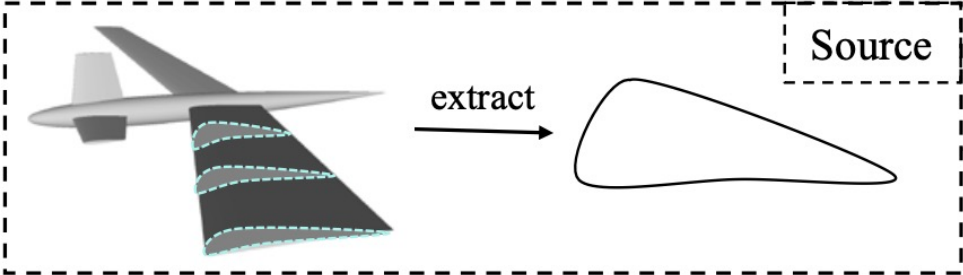
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Motivation

- The existing airfoil datasets are relatively **small-scale**.
- The current datasets typically provide only **a single condition**.
- The current methods do not support progressive **editing** existing designs.
- Lack of a **comprehensive and clear codebase** to compare and analyze different approaches.

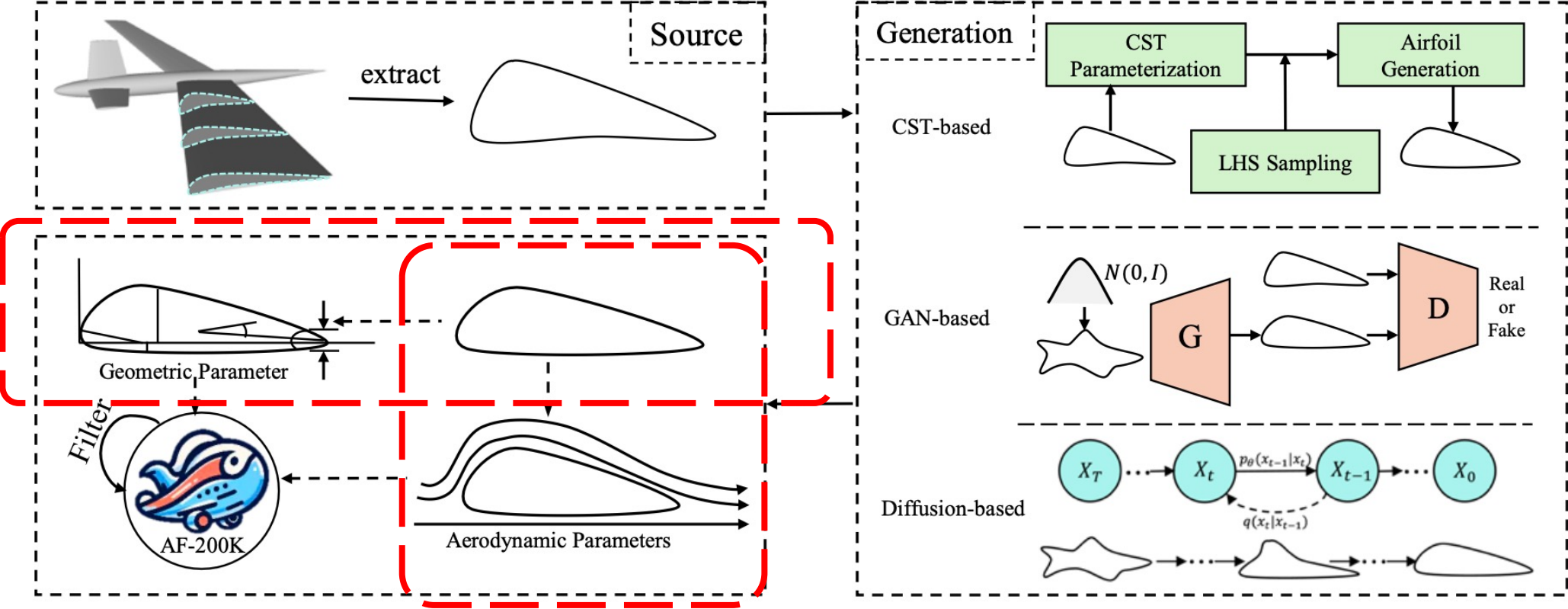
Automatic Data Engine



1. Generation stage:

- CST-assisted Generation
- Unconditional Airfoil Generation

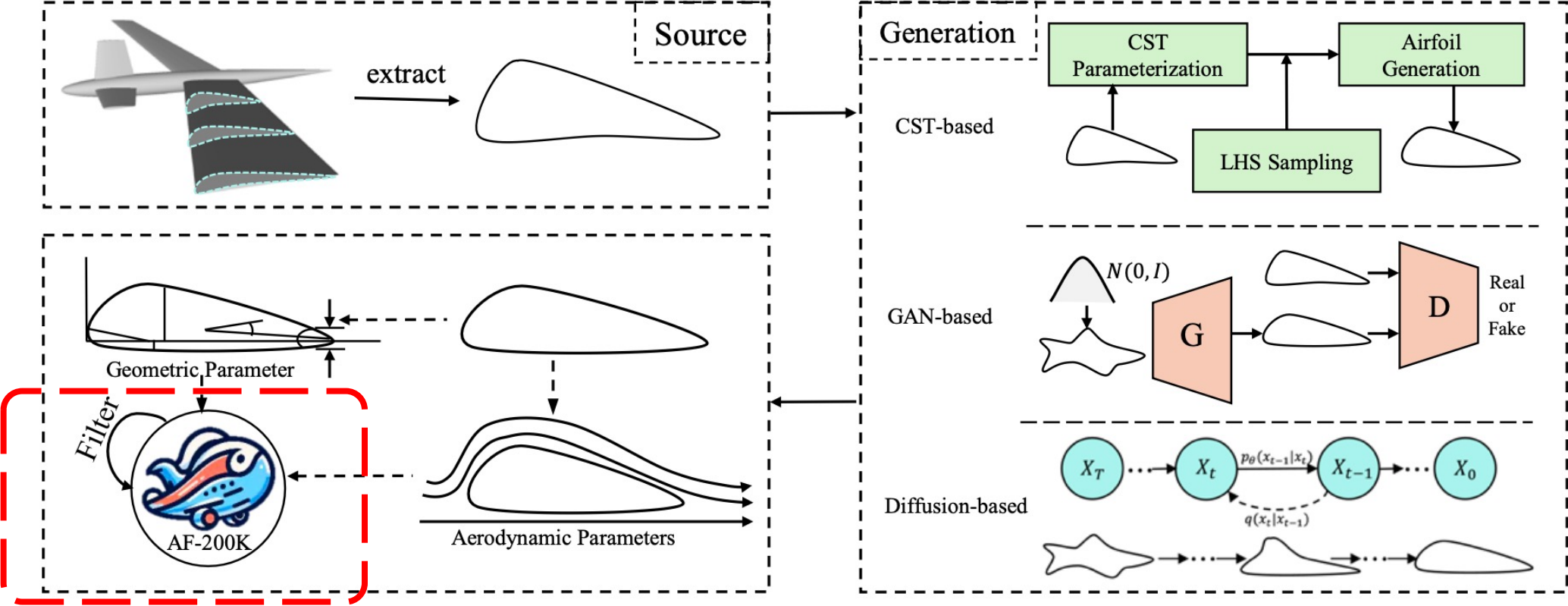
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2. Annotation stage:

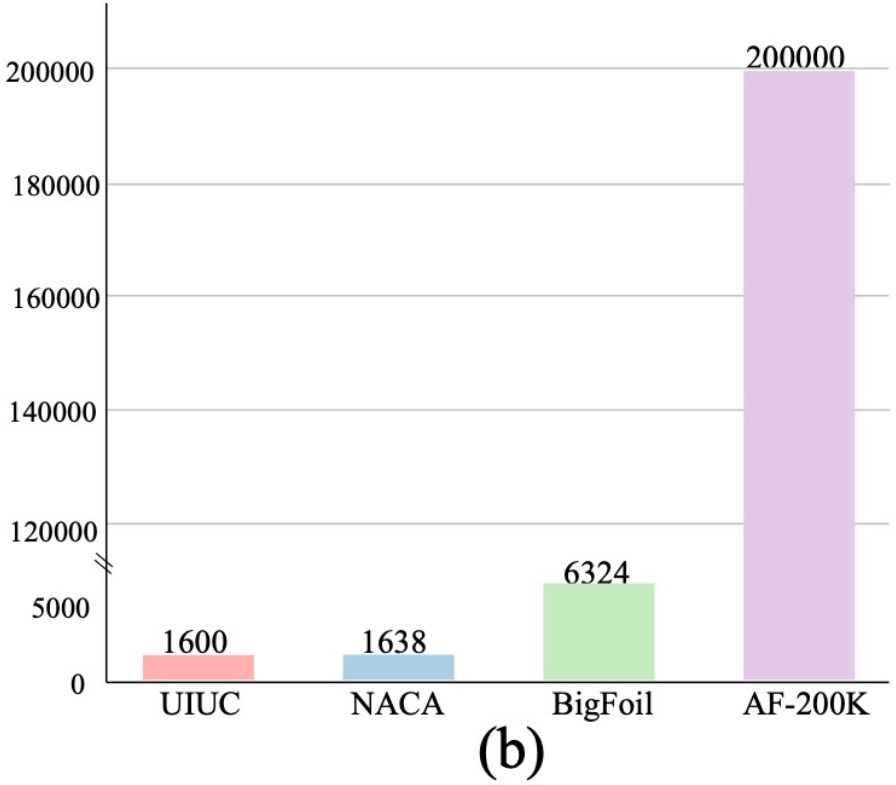
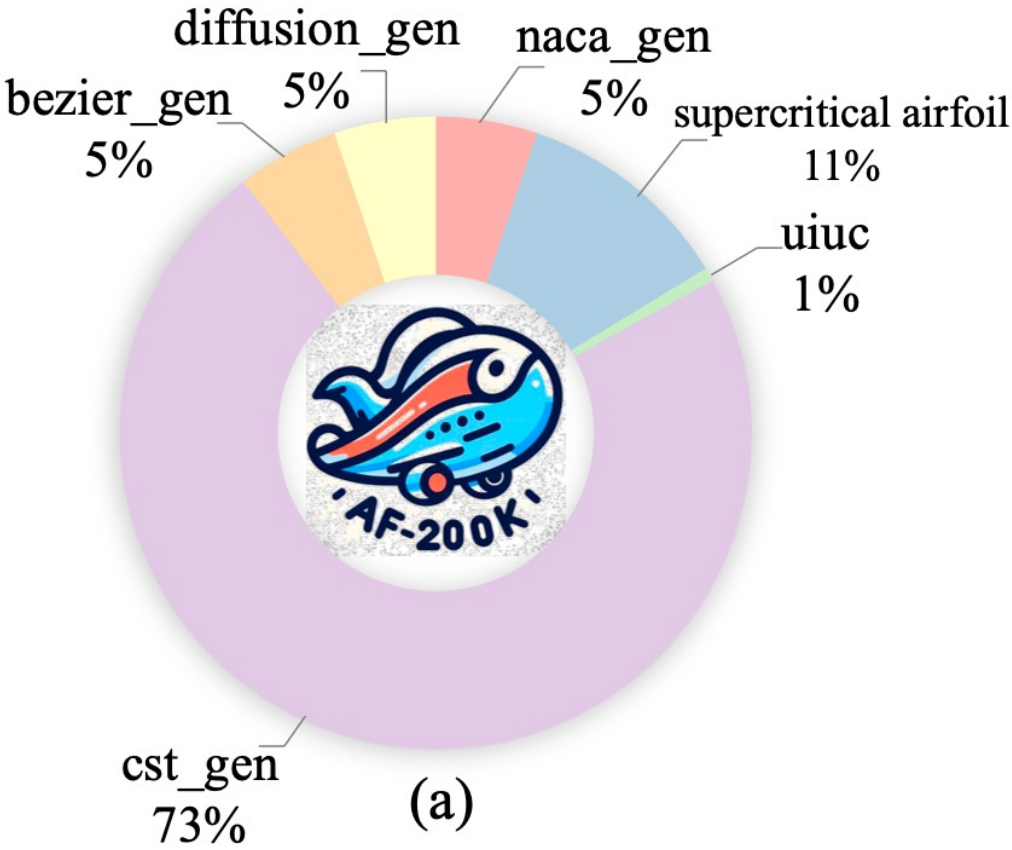
- Aerodynamic
- Geometric

Automatic Data Engine



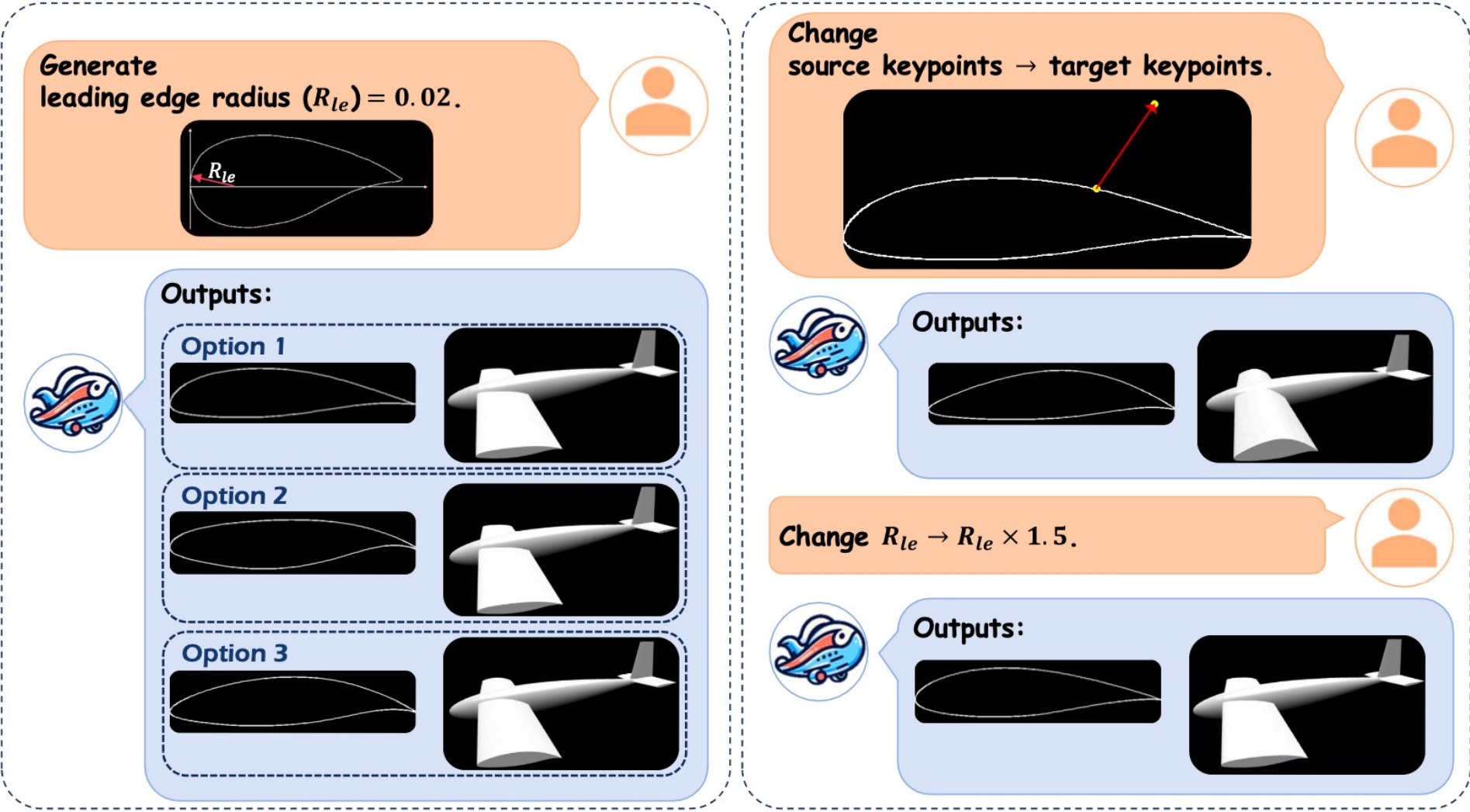
3. Filtering stage

Dataset & Benchmark



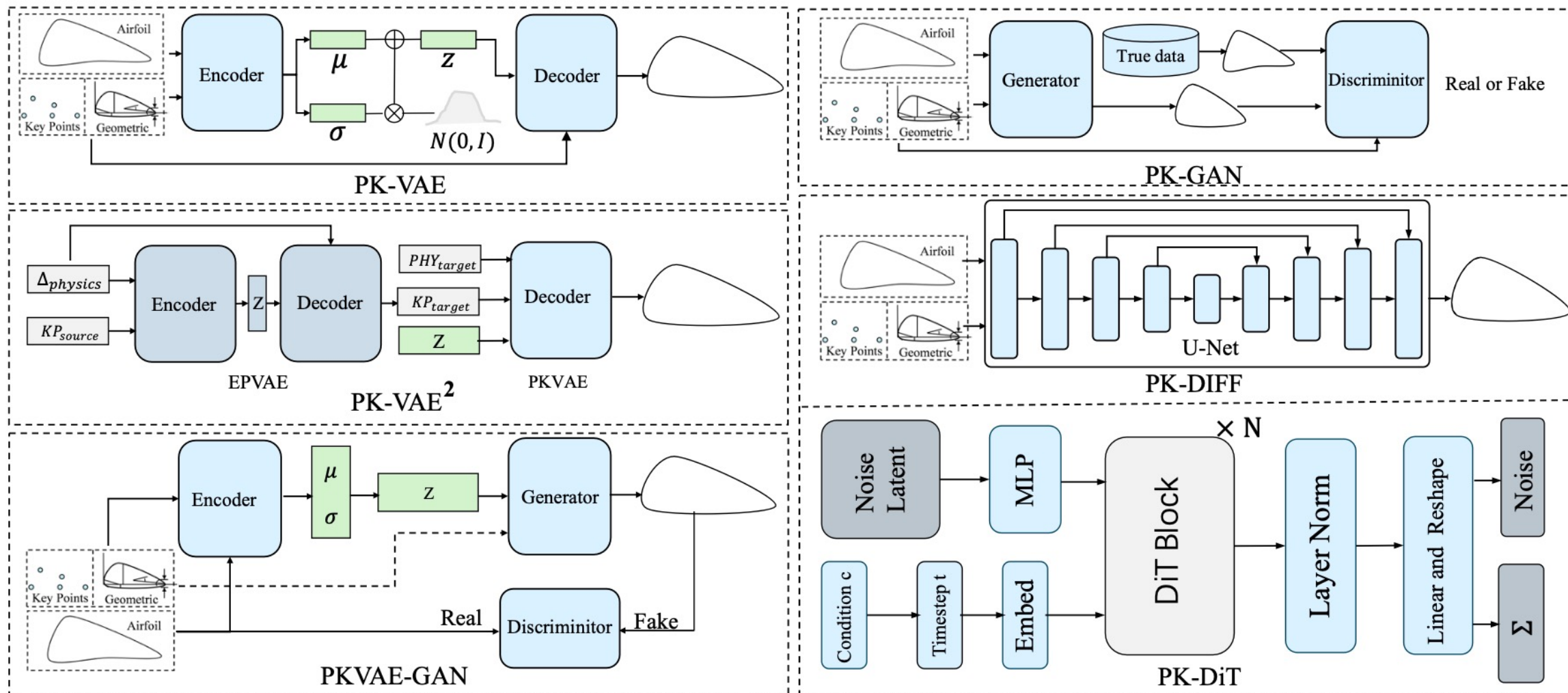
Dataset presentation

Dataset & Benchmark



Two Airfoil Inverse Design Tasks

Dataset & Benchmark



The baseline methods for benchmarking the dataset

Dataset & Benchmark

Comprehensive metrics:

- Label error: $\sigma_i = |\hat{p}_i - p_i|, i = 1, 2, \dots, 11$

- Diversity: $\mathcal{D} = \frac{1}{n} \sum_{i=1}^n \log \det(\mathcal{L}_{S_i}),$

- Smoothness: $\mathcal{M} = \sum_{i=1}^N \text{Distance}_{P_n \perp |P_{n-1} P_{n+1}|},$

- Success rate: $\mathcal{R} = \frac{1}{N} \sum_{i=1}^N \mathbb{I}\left(\frac{\sum_{j=1}^M C_j}{M} > 60\%\right), j = 1, \dots, M,$

Analysis

Method	Dataset	Label error $\downarrow \times 0.01$												$\mathcal{D} \uparrow$	$\mathcal{M} \downarrow \times 0.01$
		σ_1	σ_2	σ_3	σ_4	σ_5	σ_6	σ_7	σ_8	σ_9	σ_{10}	σ_{11}	$\bar{\sigma}$		
CVAE [16]	AF-200K	7.29	5.25	3.52	1590	9.9	9.55	2900	1.91	1.53	4.6	10.4	413.1	-155.4	7.09
CGAN [15]	AF-200K	10.7	8.50	5.44	2320	14.3	13.7	5960	2.53	2.23	5.3	12.9	759.6	-120.5	7.31
PK-VAE	AF-200K	6.30	4.79	3.13	862	6.6	6.41	1710	1.35	0.93	3.3	7.8	237.5	-150.1	5.93
PK-GAN	AF-200K	8.18	6.30	4.70	2103	12.0	11.7	3247	2.25	1.96	5.0	12.7	492.3	-112.3	3.98
PKVAE-GAN	AF-200K	5.68	3.17	3.10	565	4.6	4.35	1200	0.91	0.51	2.8	6.3	163.3	-129.6	2.89
PK-DIFF	AF-200K	4.61	3.46	2.15	277	2.2	1.93	1030	0.70	0.11	2.4	3.1	120.6	-101.3	1.52
PK-DIT	UIUC	6.38	5.14	3.36	1183	8.7	8.49	2570	1.69	1.19	3.6	9.8	345.6	-141.7	6.03
PK-DIT	Super	5.20	3.50	2.40	301	2.9	3.32	1050	0.83	0.26	2.7	3.3	125.0	-123.4	1.97
PK-DIT	AF-200K	1.12	3.23	1.54	105	1.3	1.15	979	0.05	0.05	2.3	2.4	99.7	-93.2	1.04

Performance across different datasets

Analysis

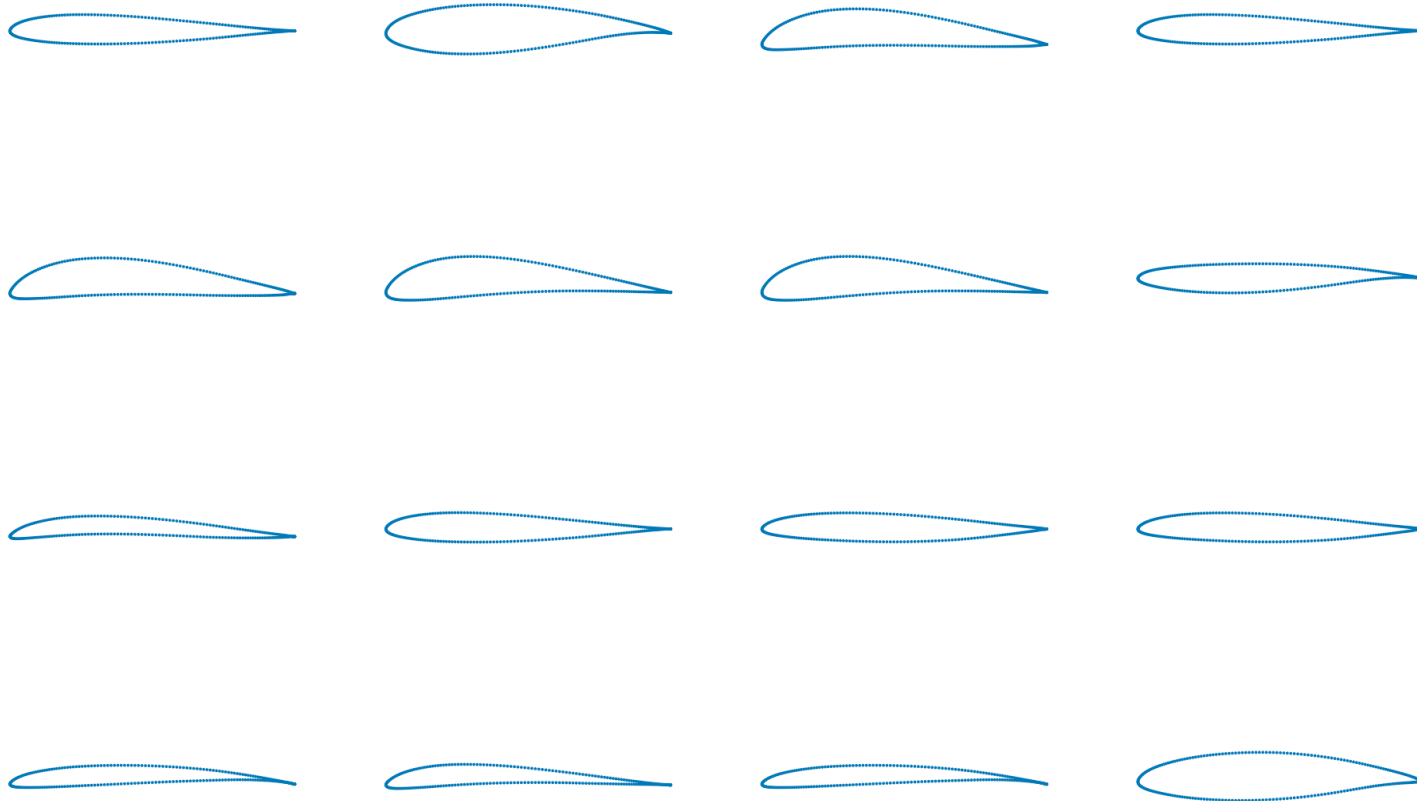
Method	Task	Label error $\downarrow \times 0.01$												$\mathcal{D} \uparrow$	$\mathcal{M} \downarrow \times 0.01$
		σ_1	σ_2	σ_3	σ_4	σ_5	σ_6	σ_7	σ_8	σ_9	σ_{10}	σ_{11}	$\bar{\sigma}$		
PK-VAE	EK	9.3	8.33	5.27	2082	12.9	11.1	4620	2.51	2.04	5.1	11.8	615.5	-143.4	7.21
PK-VAE	EP	8.9	6.38	4.94	1780	10.9	9.4	4570	2.05	1.98	4.9	10.3	582.6	-150.8	7.19
PK-VAE ²	EK	7.1	5.71	4.05	1430	8.0	8.1	3780	1.91	1.52	3.6	8.7	478.1	-133.4	6.20
PK-VAE ²	EP	6.5	5.22	3.57	1010	7.8	7.3	2010	1.52	1.03	3.4	7.9	278.5	-135.6	6.36

Performance across different design tasks

Method	Label Error ($\times 0.01$) \downarrow												$\mathcal{D} \uparrow$	$\mathcal{M} \downarrow \times 0.01$
	σ_1	σ_2	σ_3	σ_4	σ_5	σ_6	σ_7	σ_8	σ_9	σ_{10}	σ_{11}	$\bar{\sigma}$		
NACA-GEN	6.26	5.10	3.29	961	7.69	7.46	2130	1.08	1.038	3.4	8.0	284.9	-136.4	5.09
CST-GEN	5.82	4.09	2.80	572	4.61	4.36	1390	0.94	0.542	3.1	5.9	181.3	-101.5	2.31
BézierGAN-GEN	5.96	4.96	3.07	839	5.64	6.38	1900	0.98	0.817	3.1	7.4	252.5	-125.3	1.21
Diffusion-GEN	5.44	3.83	2.58	353	3.09	3.33	1180	0.89	0.293	2.9	4.2	141.8	-111.9	2.05

Performance across different generated data

Visualization

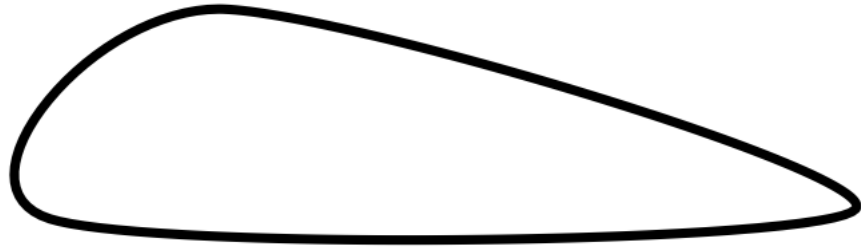


- A **large-scale and diverse** airfoil dataset.
- Improving the **capabilities** of data-driven models.
- Providing a **more comprehensive** benchmark.

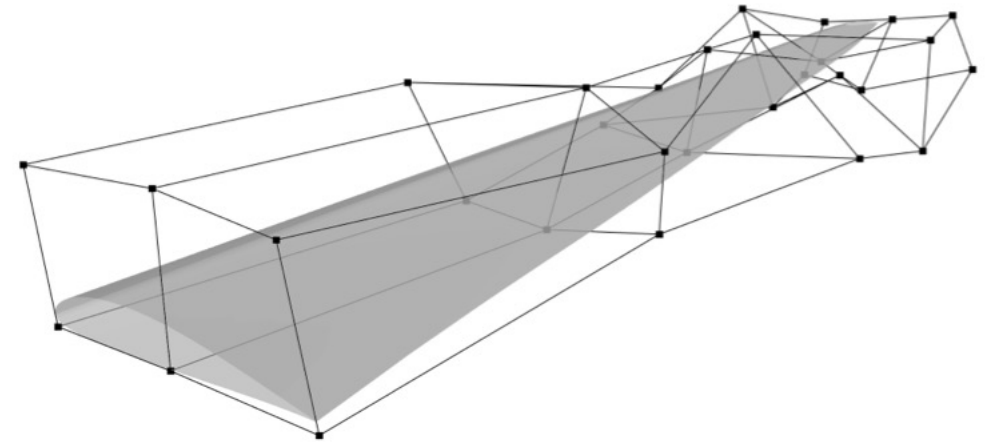
Diverse airfoils generated by PK-DIT

Limitation & Future works

- Dealing with multiple conditions
- Optimization techniques integration
- Dimension extension



2D airfoil inverse design



3D airfoil inverse design



**Deep Generative Model for Efficient 3D Airfoil
Parameterization and Generation**

Wei Chen and Arun Ramamurthy

Arxiv, 2021



Github



Zhihu



Arxiv

Thank you!